

# **Greener by Design**

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Graf-Zeppelin-Haus, Friedrichshafen  
30 June – 3 July 2003**

**The Air Travel - Greener by Design Initiative  
(Launched March 2000)**

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- **Sub-Groups:** Operations  
Technology  
Market-Based Options

# Technology Sub-Group

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Scope

In:

Noise  
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New paper:

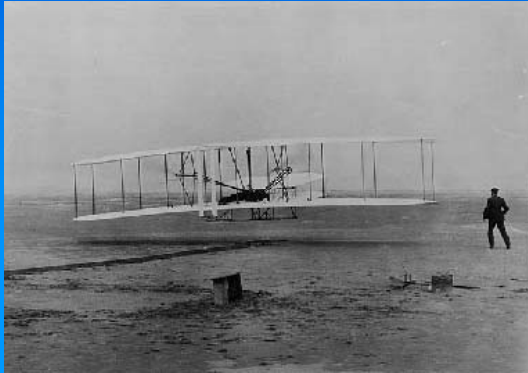
Aeronautical Journal June 2003

# Technology perspective 2 years on

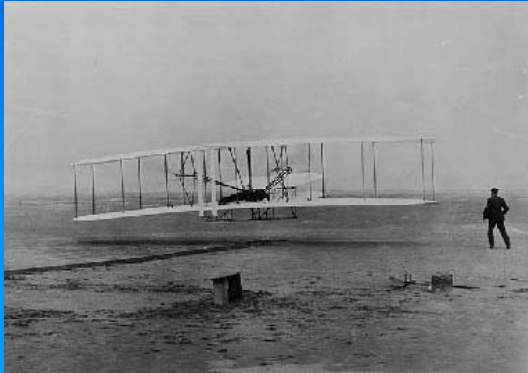
- regulation and economic instruments
- conflicts and trade-offs
- focus on climate change
  - main contributors
  - challenges to technology
    - reducing contrails
    - reducing NO<sub>x</sub>
    - reducing CO<sub>2</sub>
- design questions
- conclusions and recommendations

# Emergence of the dominant configuration

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# Emergence of the dominant configuration



- Highly evolved
- Strictly limited scope for improvement
- Commercial forces alone unlikely to break the mould

# Regulation and economic instruments



# Regulation and economic instruments

Noise

ICAO Annex 16 Chapters 3 & 4 (2006)

Local (eg Heathrow, Washington)

LAQ

ICAO CAEP/2 & CAEP/4 (2004)

Local (eg Zurich)

# Regulation and economic instruments

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Climate Change	Kyoto (excludes international flights) ICAO ) EU ) considering options HMG )

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Climate Change	Kyoto (excludes international flights) ICAO ) EU ) considering options HMG )

- climate proposals tend to be focussed on CO<sub>2</sub> emissions (with factor of 2.7 or 3 multiplier): this is likely to prove counter-productive

**Annual external costs of UK civil aviation  
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**“Aviation’s principal externality, which can be translated into monetary terms, arises from the effect of greenhouse gases and the impact they have on climate change”**

# Conflicts and trade-offs

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- on modern engines, reducing noise increases fuel burn, CO<sub>2</sub> emissions and costs



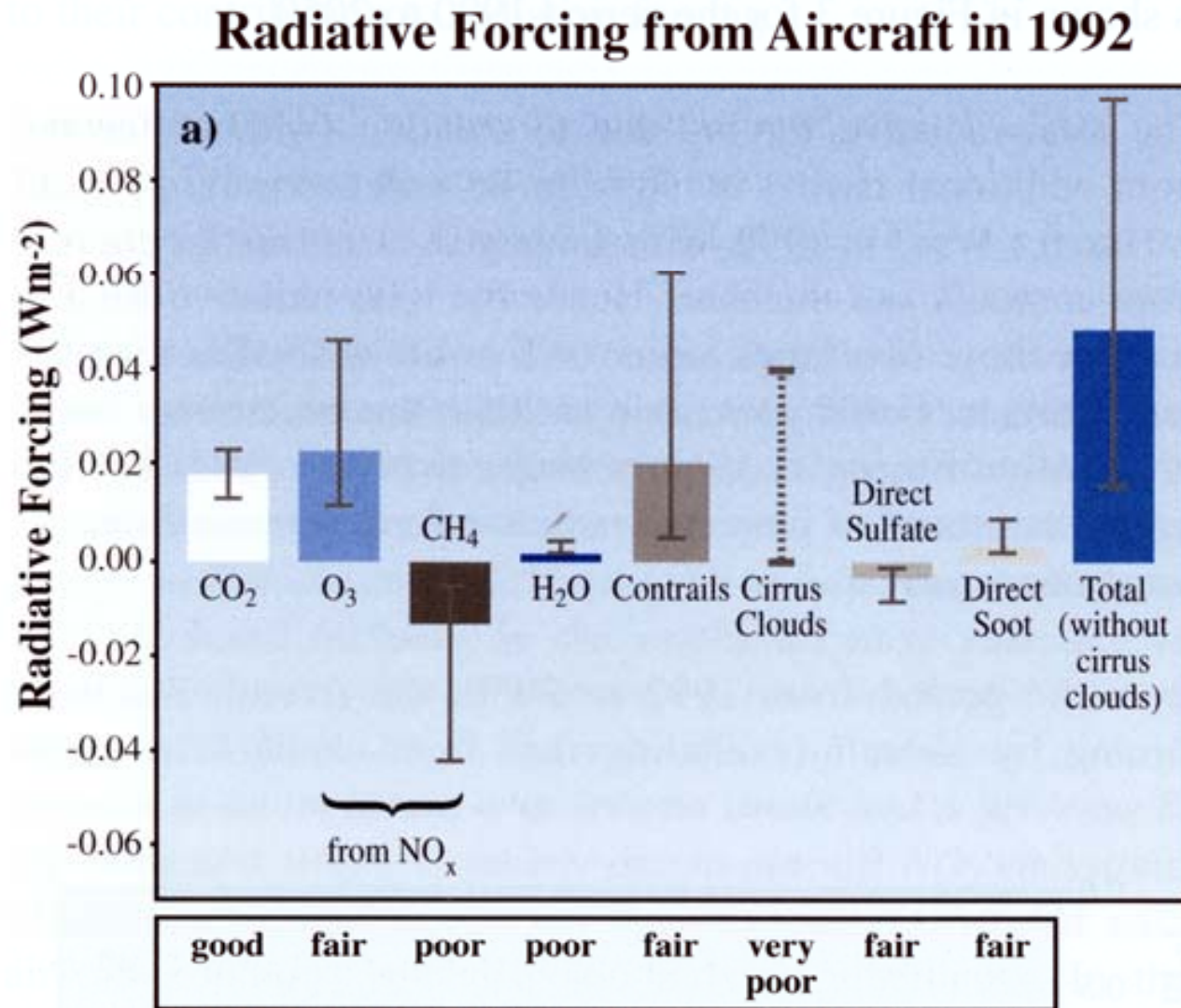
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- reducing fuel burn and CO<sub>2</sub> emissions by increasing engine thermal efficiency increases NO<sub>x</sub>
- operational measures to reduce contrails and cirrus cloud would increase fuel burn and CO<sub>2</sub> emissions

# Contributions of aviation to climate change



# Lifetimes of greenhouse gases and aircraft emissions

Carbon Dioxide	50 – 100 years
Methane	8 – 10 years
Water	days (sea level) weeks (tropopause)
Ozone	week (sea level) months (topopause)
NO <sub>x</sub>	days (sea level) weeks (tropopause)

# Challenges to technology

- Reducing persistent contrails and cirrus cloud
- Reducing impact of  $\text{NO}_x$
- Reducing  $\text{CO}_2$

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- **Persistent contrails form only in air which is saturated with respect to ice: the conditions for formation and persistence are reasonably well understood**
- **There is no prospect of preventing contrail formation in an ice-saturated atmosphere by technological means**
- **Increasing propulsive efficiency reduces the mean exhaust temperature and increases the altitude range over which contrails will form**

## **Challenges to technology: reducing persistent contrails and cirrus cloud**

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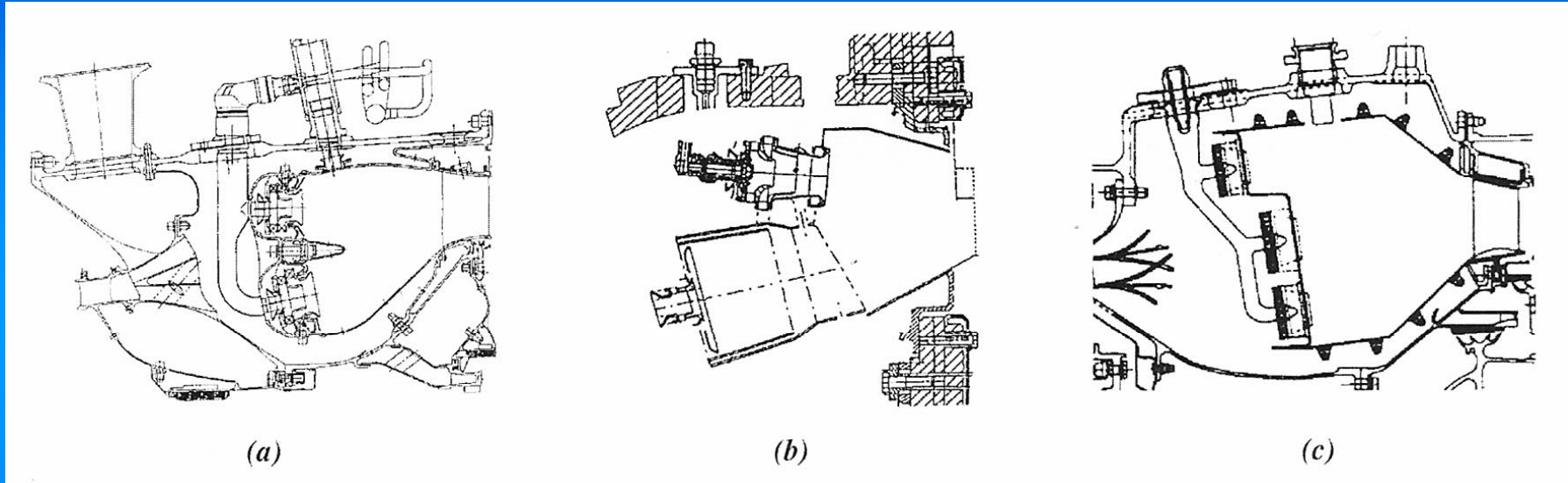
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## **Challenges to technology: reducing persistent contrails and cirrus cloud**

- Persistent contrails can be avoided by flying above, below or around ice-saturated regions: this will increase fuel burn and CO<sub>2</sub> emissions
- To minimise the economic penalty of such a strategy, future aircraft design should aim for flexibility in economic cruise altitude
- Further advances in atmospheric science, air traffic management and meteorology are needed before such a strategy can be justified or adopted
- Nevertheless, reducing persistent contrails might prove to be the single most powerful means of reducing the impact of aviation on climate, even though it would increase CO<sub>2</sub> emissions

# Challenges to technology: reducing NOX

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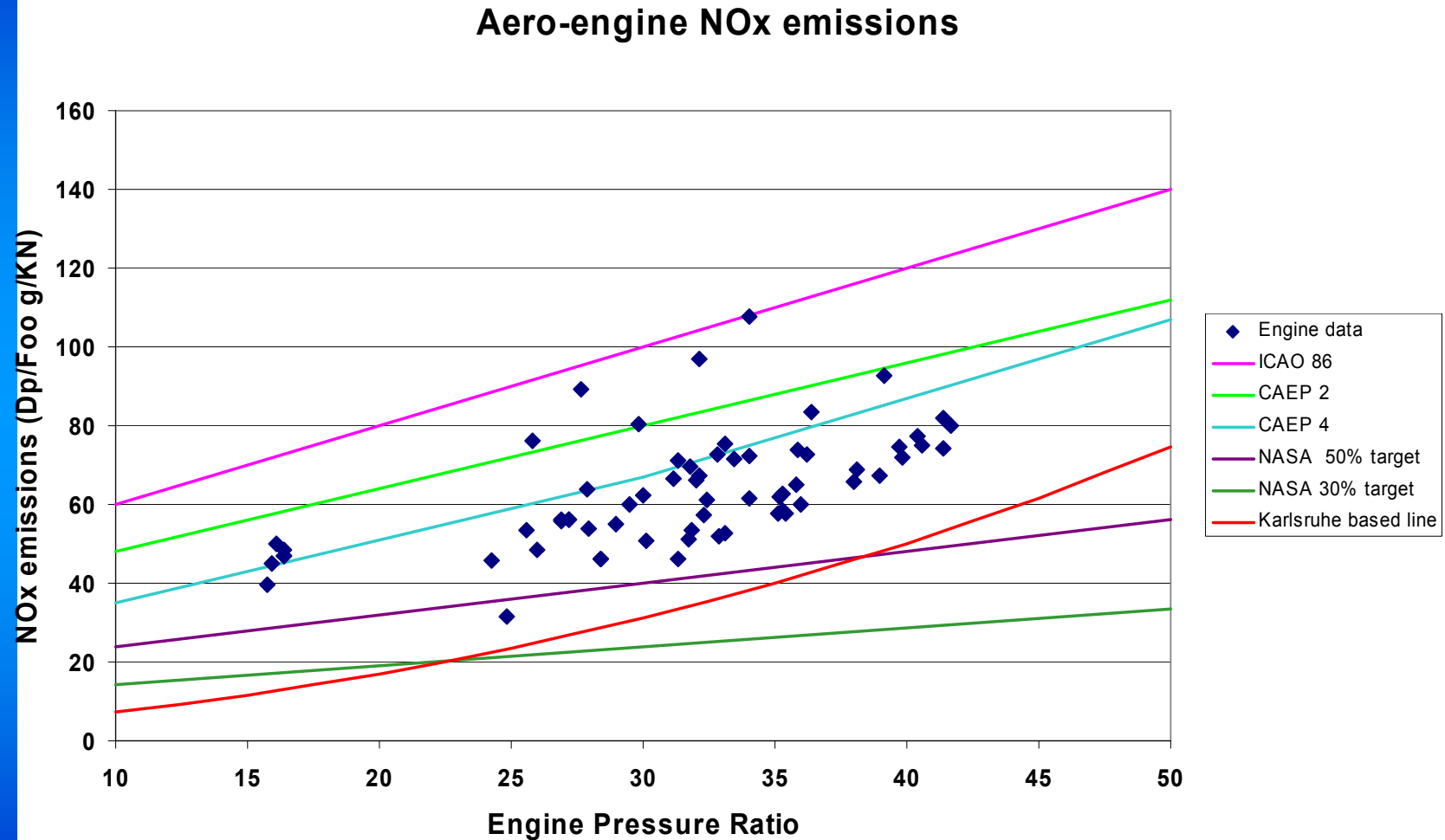
General Electric

Snecma

Pratt and Whitney

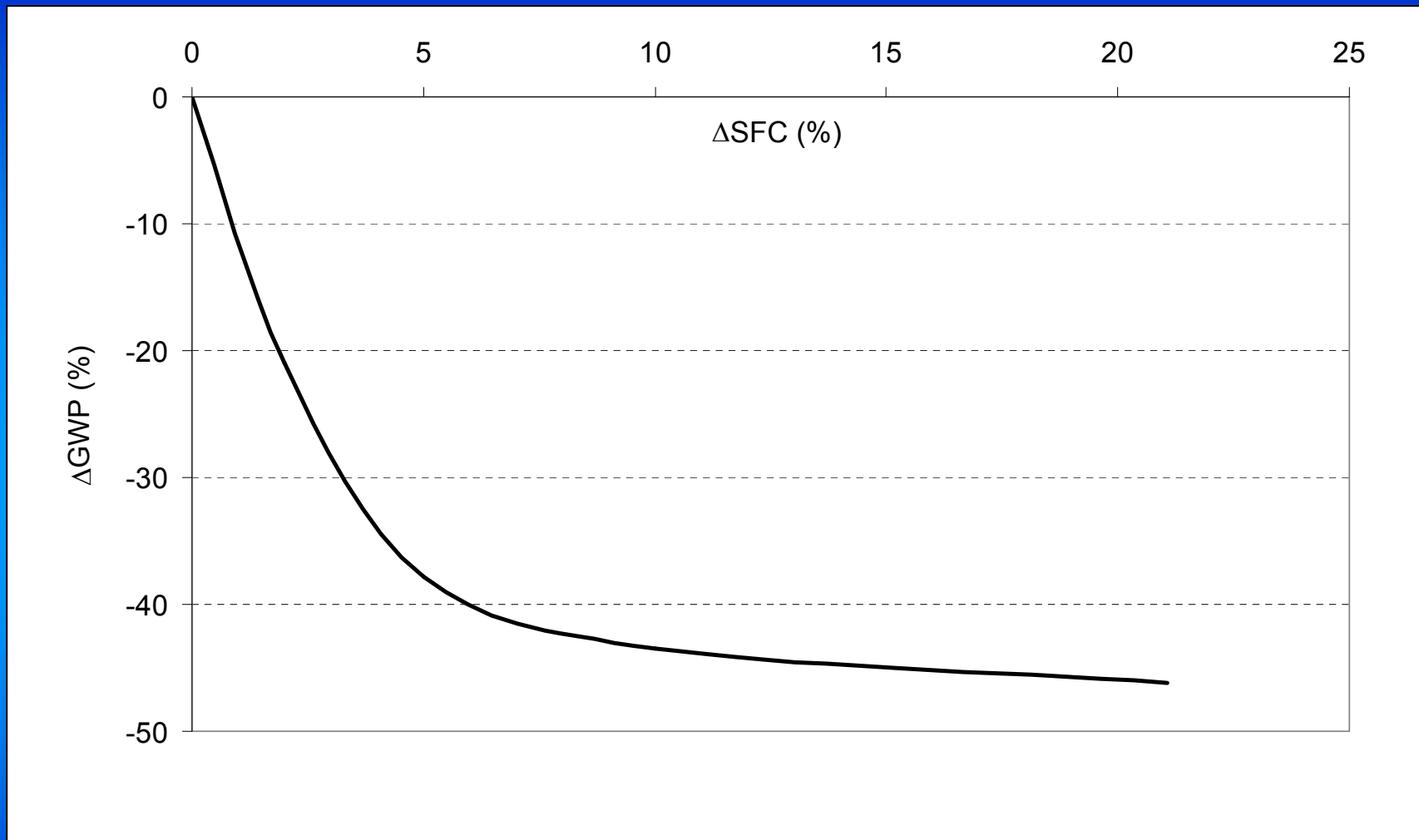
## Staged Combustors

# Challenges to technology: reducing NO<sub>x</sub>





# Challenges to technology: reducing NOX



Trade off between reduced Global Warming Potential and increased SFC relative to minimum SFC datum (Whellens and Singh)

**Challenges to technology: reducing CO<sub>2</sub>  
= reducing fuel burn per passenger km**

## Challenges to technology: reducing CO<sub>2</sub>

Fuel burn per passenger kilometre: -

$$\frac{W_f}{RW_p} = \left( 1 + \frac{W_E}{W_p} \right) \left( \frac{\exp\left(\frac{R}{X}\right) - 1}{R} \right)$$

Breguet range equation

where

X	=	HηL/D
H	=	calorific value of fuel
η	=	overall propulsive efficiency
L/D	=	lift/drag ratio

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- **Design parameters – design range, cruise Mach number**



# Challenges to technology: reducing CO<sub>2</sub> by increasing propulsive efficiency

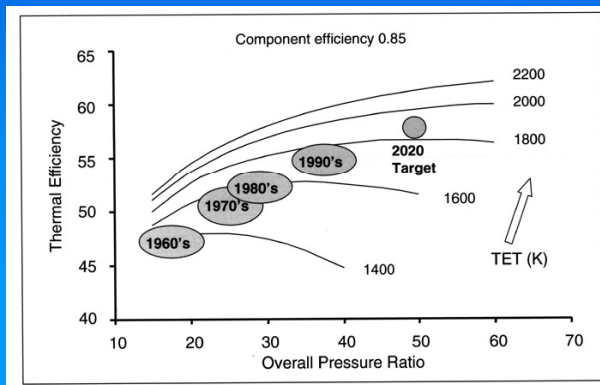
Overall propulsive efficiency

$$\eta = \eta_E \eta_P$$

where  $\eta_E$  = thermal efficiency

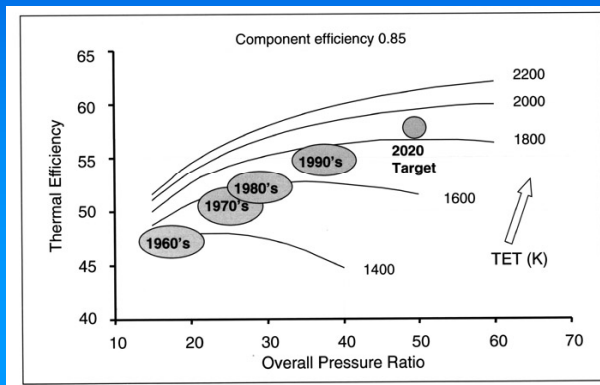
and  $\eta_P$  = jet propulsive efficiency

# Challenges to technology: reducing fuel burn by increasing propulsive efficiency - Joule cycle turbofan

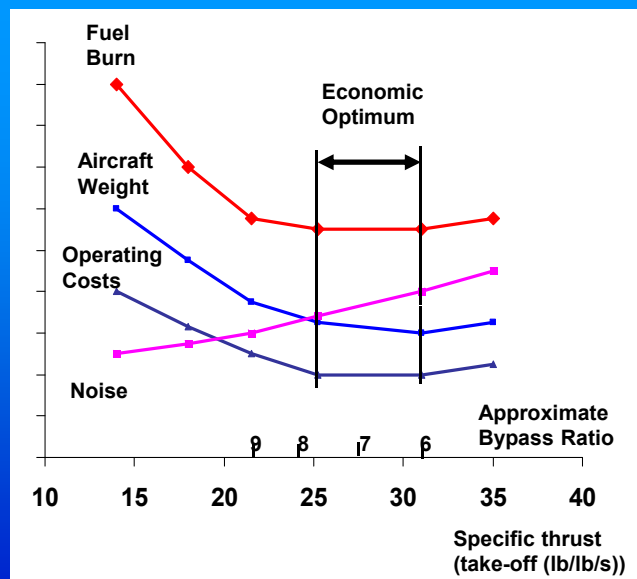


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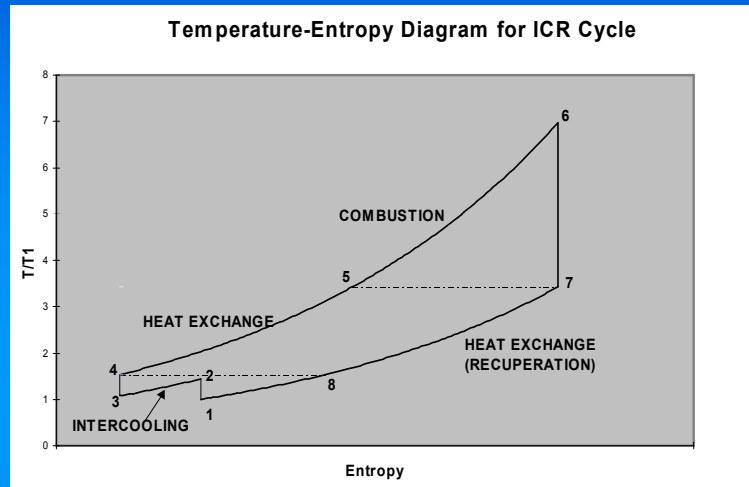


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- Most large turbofans have specific thrust around the optimum for fuel burn: reducing specific thrust below this optimum in order to meet noise targets increases fuel burn and  $\text{CO}_2$

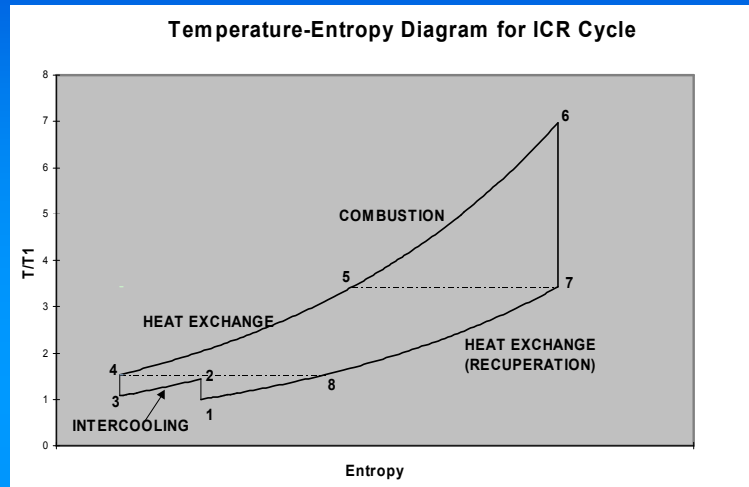
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## Intercooled recuperative engine cycle

- reduced fuel burn & CO<sub>2</sub>
- reduced NO<sub>x</sub>
- capable of podded installation
- increased weight and complexity

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## Unducted fan

- reduced fuel burn & CO<sub>2</sub>
- reduced cruise Mach number
- complexity and flight safety issues

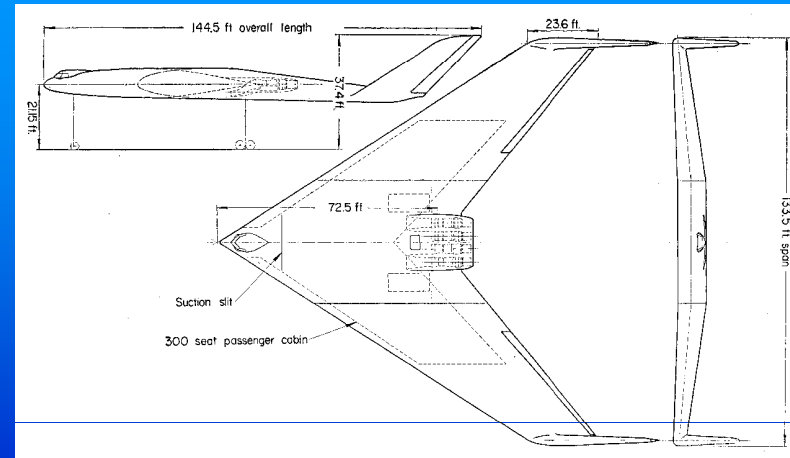
# Challenges to technology: reducing CO<sub>2</sub> by reducing drag



- Dominant configuration with hybrid laminar flow control
- Blended wing body
- All laminar flying wing

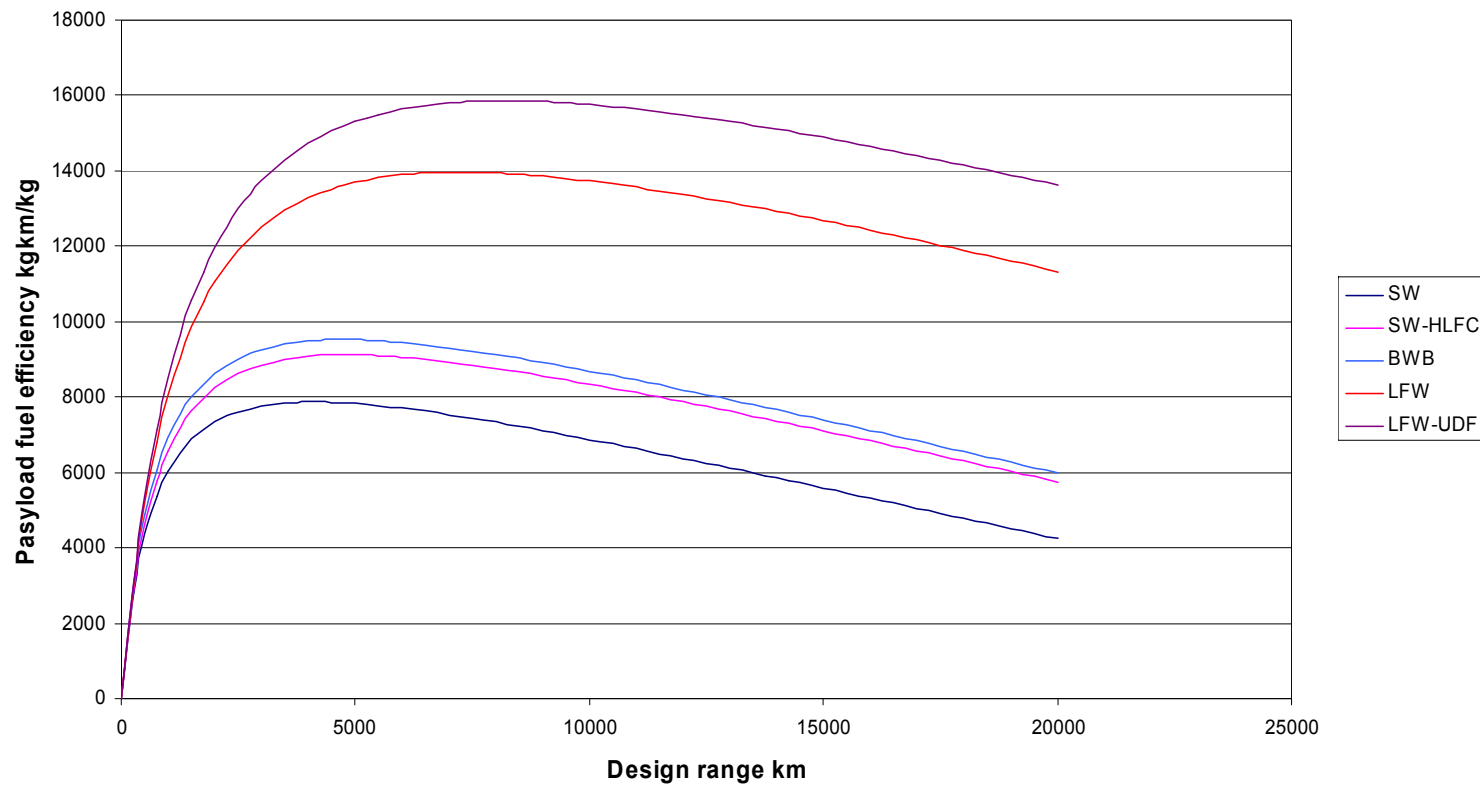


 Blended wing body model  
NASA Langley Research Center 2/20/1998 Image # EL-1998-00245



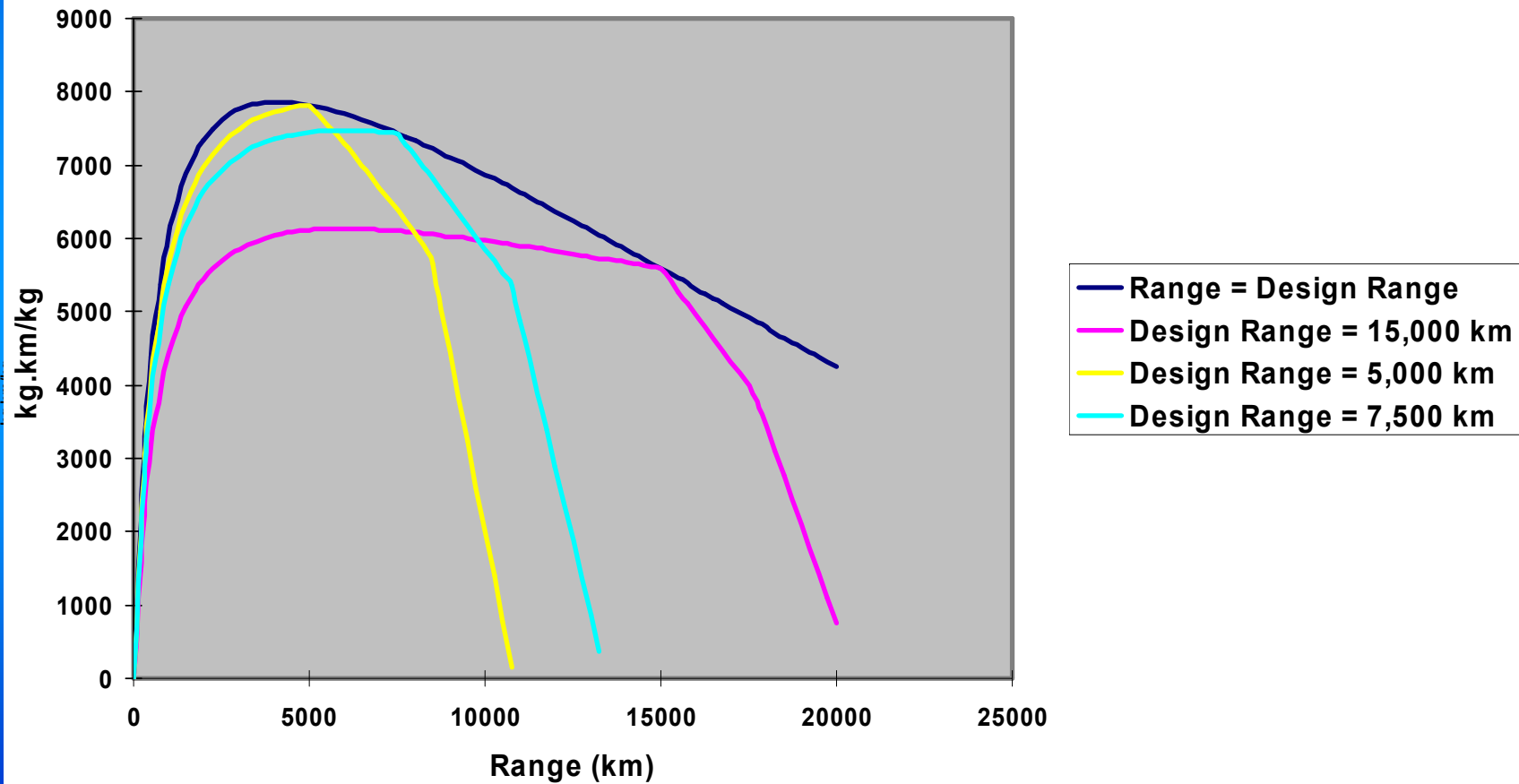
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Payload fuel efficiency versus design range for kerosene fuelled aircraft



# Challenges to technology: reducing fuel burn – effect of range

## Payload Fuel Efficiency versus Range and Design Range





## Reducing fuel burn: effect of design range

Design range km	Payload tonne	Fuel tonne	Max TOW tonne	Empty Weight tonne	Fuel for 15,000 km tonne
15,000	44.8	120.4	300.0	134.8	120.4
5,000	44.8	28.6	169.0	95.6	85.8

Multi-Sector Long Distance Travel ?

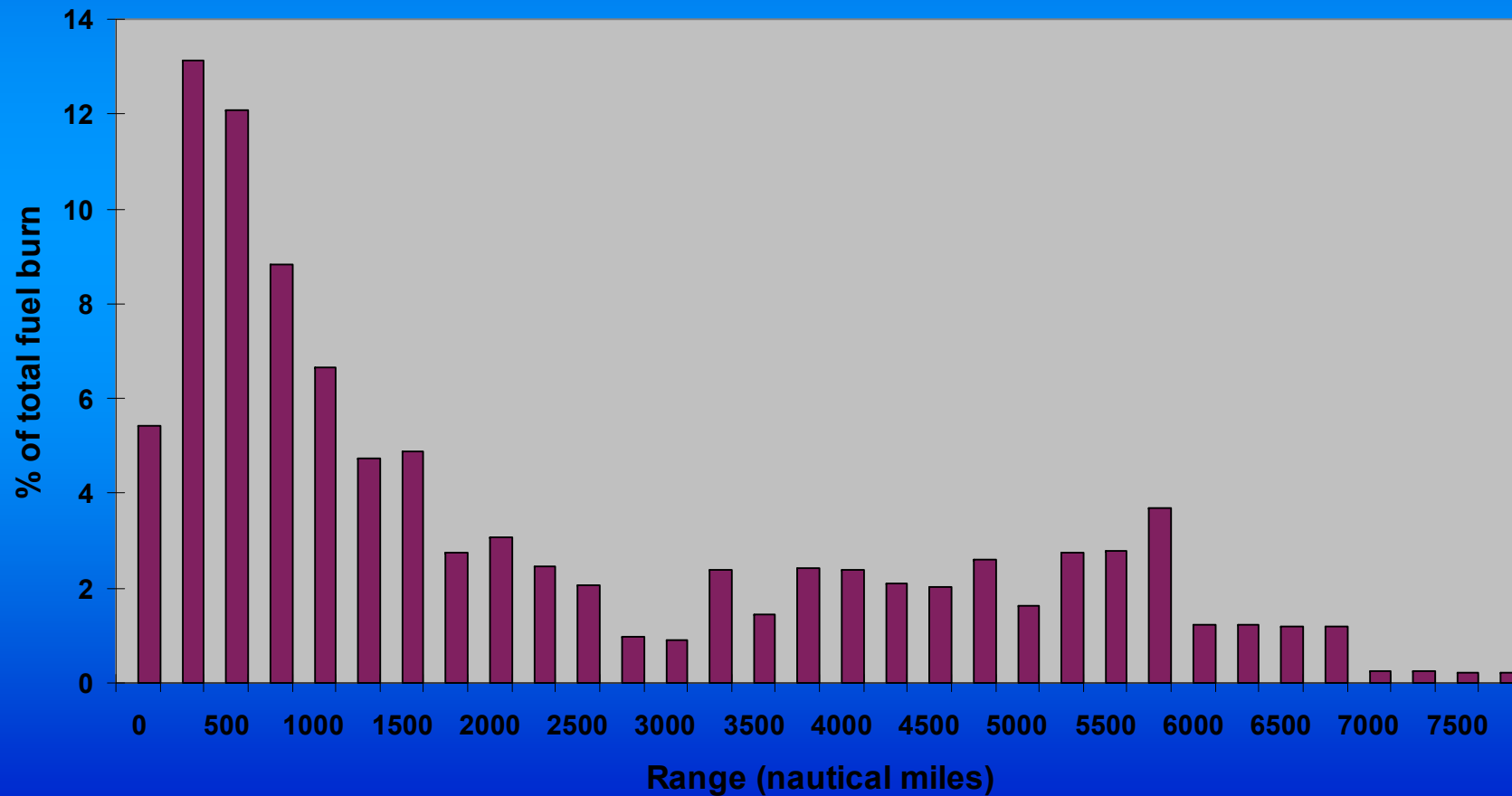
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# Significance of range

Distribution of Fuel Burn over Range  
1998 Scheduled Flights



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- **Design for minimum impact on climate**
  - trade off between operating and environmental costs ?

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# CONCLUSIONS I

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- Reducing NO<sub>x</sub> and persistent contrails are probably the two most potent means of reducing this impact: in each case, the best environmental result is likely to entail some increase in CO<sub>2</sub> emissions.
- Because CO<sub>2</sub> is such a long lived greenhouse gas, reducing its emission is a key long-term goal: drag and weight reduction are the two most potent technologies. Aircraft design parameters – design range, cruise Mach number and altitude – are also significant factors.

# CONCLUSIONS II

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# CONCLUSIONS II

- To achieve large reductions in CO<sub>2</sub> requires radical changes - a departure from the dominant configuration and the use of laminar flow control as a minimum.
- Regulatory and economic measures should be framed so as to promote the greatest possible reduction in impact on climate: measures based solely on CO<sub>2</sub> emission will probably do more harm than good.
- The challenge to technology is severe: the atmospheric science is not yet robust: the timescales for introducing new technology and new design concepts are long: the need for research and demonstration is urgent.

# **RECOMMENDATIONS – The Next Steps**

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## Design studies

- Design to minimise impact on climate
- Design to increase cruise altitude flexibility
- Multi segment long-range travel